

Service Instruction

No. 37

R.A.I. approved (16856/DTNa/T – 05/02/90)

Data:

Date: 16.06.1989

Oggetto:

Subject: Engine mount bolt inspection

Descrizione:

Description:

In the following pages, instructions are given to carry out the inspection on the engine mount bolt.

This S.I. has to be attached to the Maintenance Manual up to new revision of it.

The current Maintenance Manual is the 'SEPTEMBER 1986' issue which replaces all previous editions.

(Ref. : S.B. 70 rev. 2 and S.B. 76 rev. 1)

Applicabilità:

Applicable on: P 68 series

Variatione di peso e centrimento:

Weight and balance change: //

Documentazione allegata:

Publications attached: //

Instructions:

Insert this page adjacent to page B-27 (issued on september 1986) of the Maintenance Manual.

Add the following Special Inspection to par. (d):

- G. Every 2000 flight hours (or at least every 4 years or any time the engine is removed)

PARTENAVIA**Service Instruction N 37**

Insert this page adjacent to page B-50 (issued on september 1986) of the Maintenance Manual.

Add the following Instructions between points No. 17 and 18:

NATURE OF INSPECTION	Inspection Time		
	100 h	200 h	SPL
17(A) 1. Remove the mount bolts from airframe one-by-one starting with the lower bolts 2. With a 10x magnifier, visually inspect the bolts for cracks, circumferential scoring, corrosion or absence of protective coating on the shank (replace if necessary). 3. Imbibe the shank of the bolts in the anticorrosive S15/9 (Sp. DTD 369-A) or equivalent. Reinstall undamaged bolts (or new one). Torque to 335 to 375 inch pounds			G